

# Geodesic equation in axially symmetric space–times

— Analytic solutions and observables —

Claus Lämmerzahl  
with Eva Hackmann, Valeria Kagramanova, and Jutta Kunz



Centre for Applied Space Technology and Microgravity (ZARM),  
University of Bremen, 28359 Bremen, Germany

The higher–genus sigma function and applications  
Edinburgh 11. – 15. 10. 2010



# Geodesic equation in axially symmetric space–times

— Analytic solutions and observables —

Claus Lämmerzahl  
with Eva Hackmann, Valeria Kagramanova, and Jutta Kunz



Centre for Applied Space Technology and Microgravity (ZARM),  
University of Bremen, 28359 Bremen, Germany

The higher–genus sigma function and **applications**  
Edinburgh 11. – 15. 10. 2010



# Where we are



# Where we are



# Where we are



# The Bremen drop tower



# The Bremen drop tower

## Space Science

- Fundamental Physics
- Key Technologies
- Control systems
- Space technology
- Micro satellites

## Fluid mechanics

- Fluid dynamics
- Energy and propulsion
- Computational fluid dynamics
- Experimental fluid mechanics



# Fundamental Physics at ZARM: Scope

## Scope of Fundamental Physics at ZARM

- **Development of new technologies**
  - for microgravity experiments (drop tower, ISS, satellite)
  - for applications in space
- Accompanying **theoretical investigations**
  - motivation for experiments and missions
  - theory for experiments and applications
- **High precision modeling**
  - experimental devices
  - whole spacecraft
  - whole missions
  - quantum modeling

23 members

3 Professors, 3 post-docs, 14 PhD students, 1 diploma student, 2 technicians



# Fundamental Physics at ZARM

Center of Applied Space Technology and Microgravity

Research areas



Center of Applied **Space Technology** and Microgravity

Research areas

- **Satellite dynamics**
  - modeling
  - disturbance forces
  - thermal and stress analysis
  - HPS (High Performance satellite dynamics Simulator)

Center of Applied Space Technology and **Microgravity**

Research areas

- **Satellite dynamics**
  - modeling
  - disturbance forces
  - thermal and stress analysis
  - HPS (High Performance satellite dynamics Simulator)
- **Quantum physics**
  - Bose–Einstein Condensate, BEC (exp, theory & modeling)
  - atom interferometry (exp & theory)
  - quantum tests (equivalence principle, decoherence, linearity, ...)
  - development of corresponding space technology

## Center of Applied Space Technology and Microgravity

### Research areas

- **Satellite dynamics**
  - modeling
  - disturbance forces
  - thermal and stress analysis
  - HPS (High Performance satellite dynamics Simulator)
- **Quantum physics**
  - Bose–Einstein Condensate, BEC (exp, theory & modeling)
  - atom interferometry (exp & theory)
  - quantum tests (equivalence principle, decoherence, linearity, ...)
  - development of corresponding **space technology**

Center of Applied Space Technology and Microgravity

Research areas

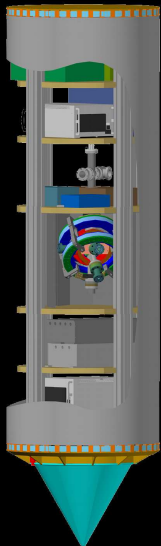
- **Satellite dynamics**
  - modeling
  - disturbance forces
  - thermal and stress analysis
  - HPS (High Performance satellite dynamics Simulator)
- **Quantum physics**
  - Bose–Einstein Condensate, BEC (exp, theory & modeling)
  - atom interferometry (exp & theory)
  - quantum tests (equivalence principle, decoherence, linearity, ...)
  - development of corresponding space technology
- **Gravitational physics**
  - tests of Equivalence Principle
  - analytical and numerical solutions for orbits
  - quantum gravity phenomenology
  - theoretical description of experiments testing SR and GR

Center of Applied Space Technology and Microgravity

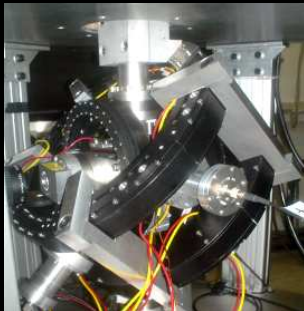
Research areas

- **Satellite dynamics**
  - modeling
  - disturbance forces
  - thermal and stress analysis
  - HPS (High Performance satellite dynamics Simulator)
- **Quantum physics**
  - Bose–Einstein Condensate, BEC (exp, theory & modeling)
  - atom interferometry (exp & theory)
  - quantum tests (equivalence principle, decoherence, linearity, ...)
  - development of corresponding space technology
- **Gravitational physics**
  - tests of Equivalence Principle
  - **analytical** and numerical solutions for orbits
  - quantum gravity phenomenology
  - theoretical description of experiments testing SR and GR

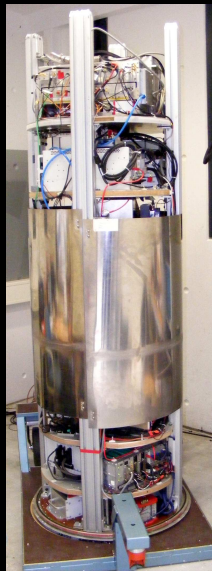
# BEC in microgravity



design of capsule

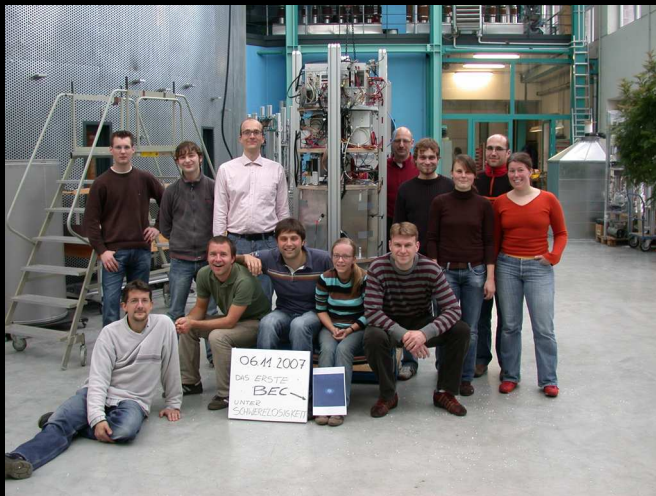


vacuum chamber



capsule

# First BEC in microgravity / extended free fall



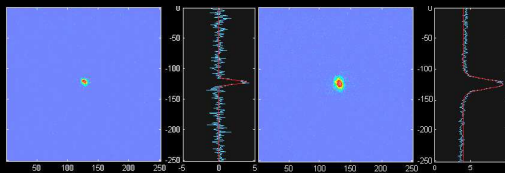
LU Hannover, ZARM, MPQ Munich, U Hamburg, HU Berlin, U Ulm





# BEC in microgravity – long free evolution

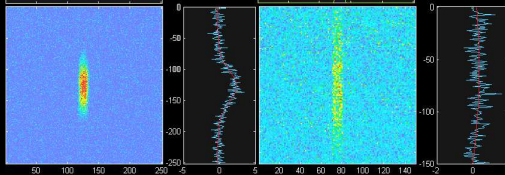
50 ms



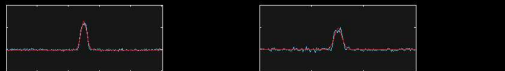
100 ms



500 ms



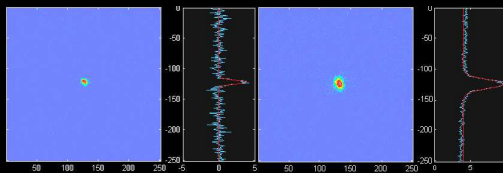
1000 ms



$10^4$  atoms, 1 s free evolution time (not possible on ground)  
van Zoest et al, Science 2010

# BEC in microgravity – long free evolution

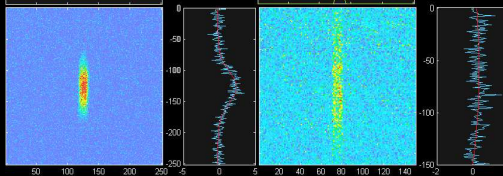
50 ms



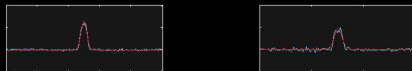
100 ms



500 ms



1000 ms



$10^4$  atoms, 1 s free evolution time (not possible on ground)  
van Zoest et al, Science 2010

# BEC in free fall

- **Status**

- until now almost 200 drops
- BEC is created regularly
- extremely robust (survives  $\sim 50 g$ )

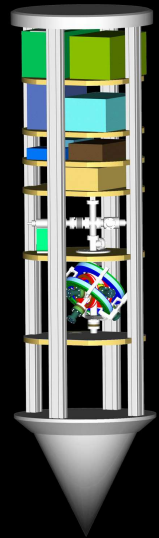
Worldwide **most advanced technology** towards space application and fundamental quantum physics in  $\mu g$

- **Ongoing work**

- PRIMUS (PRäzisions-Interferometrie mit Materiewellen Unter Schwerelosigkeit)
- FOKUS (FaserOptischer FrequenzKamm Unter Schwerelosigkeit)
- ATUS (Atom Interferometer Modeling)
- Fluctuations in Quantum Systems

- **In future**

- Fundamental Physics experiments
- Drop tower — Texus — ISS
- Inertial sensors
- High precision clocks



# High precision quantum modeling

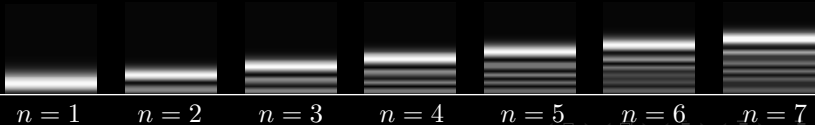
GOST: trampolin with a Bose–Einstein condensate



BEC described by Gross–Pitaevskii equation

$$i\partial_t\psi = \Delta\psi + V(\mathbf{x})\psi + g|\psi|^2\psi$$

- wide class of solutions known
- dynamical solutions in gravitational field  $V(\mathbf{x}) = g \cdot \mathbf{x}$  (Chen & Lee, PRL 1976)
- stationary and dynamical solutions in gravito–optical surface trap: boundary condition  $\psi = 0$  for  $z = 0$
- nonlinear hydrogen atom  $V(\mathbf{x}) = 1/r$
- solutions in periodic potentials
- solutions in gravitational waves
- solitons, vortices



# Multi-component BEC

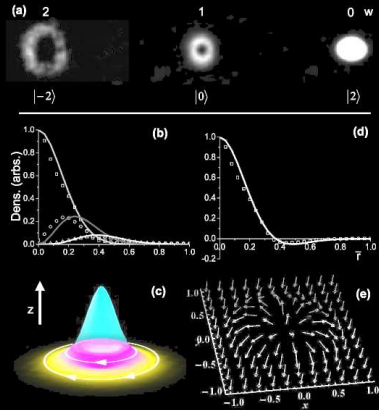
for testing the Universality of Free Fall:  
 two BECs in the same trap  
 → multicomponent BEC  
 multi-component BEC described by

$$i\partial_t \Psi_a = \Delta \Psi_a + V(\mathbf{x})\Psi_a + g\|\Psi\|^2\Psi_a$$

with

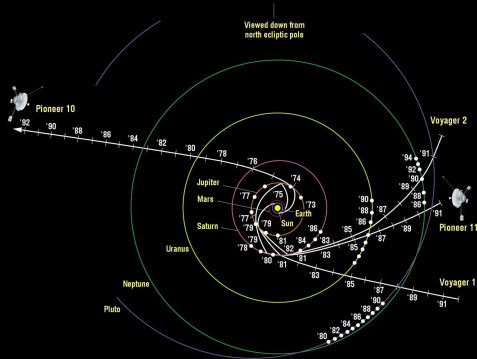
$$\Psi = \begin{pmatrix} \Psi_1 \\ \vdots \\ \Psi_n \end{pmatrix}$$

- vector Schrödinger equation:  
analytical solutions? In  
gravitational fields?
- skyrmions
- ...



Leslie et al, PRL 2009

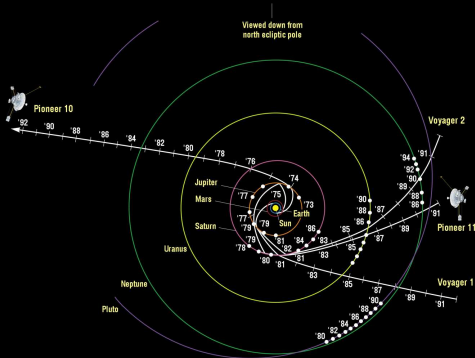
# The Pioneer anomaly



Anomalous acceleration toward Sun

- anomalous gravity?
- systematics?

# The Pioneer anomaly



Anomalous acceleration toward Sun

- anomalous gravity?
- **systematics?**



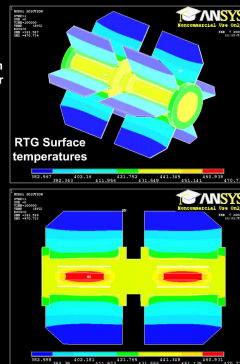
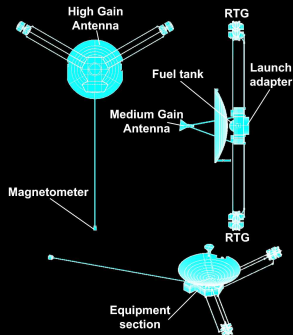
# High precision modeling

New method for high precision modeling of

- experimental devices
- spacecraft

Needed for

- ground experiments (Michelson–Morley, gravitational wave interferometers, etc.)
- analysis of Pioneer anomaly
- LISA, LISA pathfinder, geodesy missions, ...



Rievers, C.L., List, Bremer & Dittus, NJP 2009

## Main topics

- Thermal modeling
- Strains and stresses
- Drag and recoil forces
- Analytical solutions

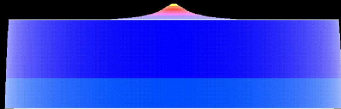
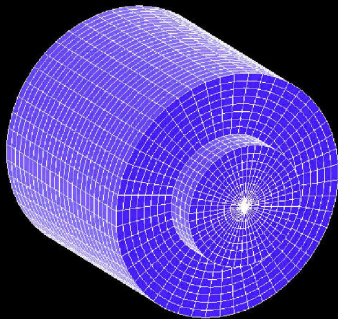
## Aim

- Accuracy  $\sim 10^{-20}$
- motivated by cavities and optical clocks

$$\nu = \frac{2n\pi}{L}c \quad \Rightarrow \quad \frac{\delta c}{c} = \sqrt{\left(\frac{\delta \nu}{\nu}\right)^2 + \left(\frac{\delta L}{L}\right)^2}$$

future frequency stability  $\sim 10^{-18}$  requires same mechanical stability

## Modeling an optical resonator



applications to gravitational wave interferometers, LISA, LISA Pathfinder, clock missions, ...

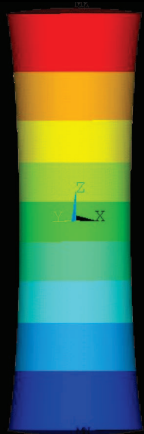
Analytic solution of Lamé-Navier equation in gravity gradient

$$\mu \Delta u^i + (\lambda + \mu)(\text{grad div } u)^i + K^i = 0$$

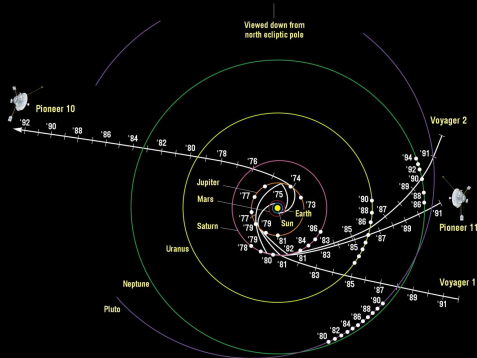
Homogenous part implies biharmonic equation

$$\Delta \Delta u = 0$$

Scheithauer & C.L., CQG 2006



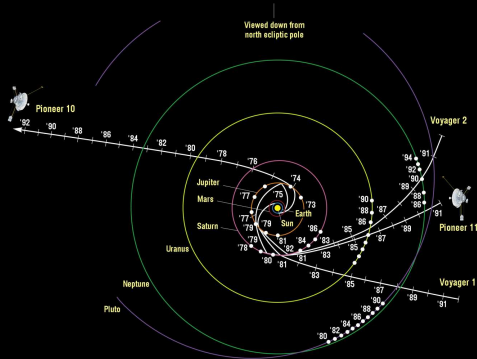
# The Pioneer anomaly



## Anomalous acceleration toward Sun

- anomalous gravity?
- systematics?

# The Pioneer anomaly



Anomalous acceleration toward Sun

- anomalous gravity?
- systematics?

# Orbits in gravitational fields

- New analytic solutions, based on algebra–geometric methods
- Analytic solution of differential equations of the form

$$\left(\frac{dx}{ds}\right)^2 = P_n(x) \quad \text{and} \quad \left(x \frac{dx}{ds}\right)^2 = P_n(x)$$

$P_n$  = polynomial of degree  $n$

- Application: influence of cosmological constant on motion
  - Plebański–Demiański space–times without acceleration in 4D (Petrov D)
  - higher dimensions
  - in space–times with mass multipoles
- Practical application
  - Pioneer anomaly, dark matter problems
  - geodesy
  - clocks in space

Hackmann & Lämmerzahl, PRL 2008, PRD 2008,  
Hackmann, Kagramanova, Kunz & Lämmerzahl, PRD 2008, 2009, 2010, EPL  
2009



# Outline

## 1 Introduction and motivation



# Outline

- 1 Introduction and motivation
- 2 General Relativity

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# The problem

gravity can only be explored through the motion of particles and light

- particles (spacecraft, stars, pulsars, black holes, ...) and light
  - point particles and light rays  $\rightarrow$  geodesic equation
  - spinning particles and polarized light  $\rightarrow$  MPD equation
  - particles with mass multipoles  $\rightarrow$  MPD equation

# Known solutions

gravity can only be explored through the motion of particles and light

- analytic solutions for geodesic equations in vacuum space-times
  - Schwarzschild (Hagihara, JJGA 1931)
  - Reissner–Nordström (Chandrasekhar 1983)
  - Kerr (Carter 1968, Chandrasekhar 1983)
  - Schwarzschild–de Sitter (Hackmann & C.L. PRL 2008, PRD 2008)
  - spherically symmetric space-times in higher dimensions (Hackmann, Kagramanova, Kunz, C.L., PRD 2008)
  - Plebański–Demiański (Hackmann, Kagramanova, Kunz, C.L., EPL 2009)
  - Kerr–de Sitter (Hackmann, Kagramanova, Kunz, C.L., PRD 2010)
  - Taub–NUT (Kagramanova, Kunz, Hackmann, C.L., PRD 2010)
  - Taub–NUT–de Sitter (Hackmann, Kagramanova, Kunz, C.L., in preparation)
- analytic solutions for geodesic equations in nonvacuum space-times
  - Schwarzschild–string (Hackmann, Hartmann, C.L., Sirimachan, PRD 2010)
  - Kerr–string (Hackmann, Hartmann, C.L., Sirimachan, PRD 2010)
  - Plebański–Demiański–string (Hackmann, Hartmann, C.L., Sirimachan, in prep.)



# Known solutions

gravity can only be explored through the motion of particles and light

- analytical solutions for extended particles
  - spinning particles in Schwarzschild (Micolaut, ZP 1967)
  - spinning particle in spherically symmetric space-times (C.L. & Schaffer, in prep.)

# Applications

- analytical calculation of satellite orbits
- analytical calculation of general relativistic effects
  - huge perihelion shift (binary black holes), Lense–Thirring effect, etc
- analytical calculation of effects of generalized gravity theories
- tests of numerical codes (for gravitational wave templates)
- binary systems and gravitational waves
  - calculation of gravitational wave templates for EMRIs
  - technique can be applied to effective one–body formalism
  - self force calculation
- accretion discs
- further application: motion in mass multipole fields
- pulsar timing formula

inclusion of spin / quadrupole  $\rightarrow$  modification, in particular enhancement, of effects

further issue: solutions of field equations



# Outline

- 1 Introduction and motivation
- 2 **General Relativity**
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# The geometry and the Einstein equations

All foundations and predictions of GR are experimentally well tested and confirmed

## Foundations

### The Einstein Equivalence Principle

- Universality of Free Fall
- Universality of Gravitational Redshift
- Local Lorentz Invariance

# The geometry and the Einstein equations

All foundations and predictions of GR are experimentally well tested and confirmed

## Foundations

### The Einstein Equivalence Principle

- Universality of Free Fall
- Universality of Gravitational Redshift
- Local Lorentz Invariance



## Implication

Gravity is a metrical theory

# The geometry and the Einstein equations

All foundations and predictions of GR are experimentally well tested and confirmed

## Foundations

### The Einstein Equivalence Principle

- Universality of Free Fall
- Universality of Gravitational Redshift
- Local Lorentz Invariance



## Implication

Gravity is a metrical theory



## Predictions for metrical theories

- Solar system effects
  - Perihelion shift
  - Gravitational redshift
  - Deflection of light
  - Gravitational time delay
  - Lense–Thirring effect
  - Schiff effect
- Strong gravitational fields
  - Binary systems
  - Black holes
- Gravitational waves

# The geometry and the Einstein equations

All foundations and predictions of GR are experimentally well tested and confirmed

## Foundations

### The Einstein Equivalence Principle

- Universality of Free Fall
- Universality of Gravitational Redshift
- Local Lorentz Invariance



## Implication

Gravity is a metrical theory



## Predictions for metrical theories

- Solar system effects
  - Perihelion shift
  - Gravitational redshift
  - Deflection of light
  - Gravitational time delay
  - Lense–Thirring effect
  - Schiff effect
- Strong gravitational fields
  - Binary systems
  - Black holes
- Gravitational waves



$$R_{\mu\nu} - \frac{1}{2}g_{\mu\nu}R = \kappa T_{\mu\nu}$$

# Equations of motion: Point particles and light rays

- Geodesic equation

$$0 = (D_u u)^\mu = \frac{d^2 x^\mu}{ds^2} + \{\rho\sigma\}^\mu \frac{dx^\rho}{ds} \frac{dx^\sigma}{ds}$$

with

$$\{\rho\sigma\}^\mu = \frac{1}{2} g^{\mu\nu} (\partial_\rho g_{\mu\sigma} + \partial_\sigma g_{\mu\rho} - \partial_\mu g_{\rho\sigma})$$

and

$$\begin{aligned} g_{\mu\nu} u^\mu u^\nu &= 1 && \text{for point particles} \\ g_{\mu\nu} u^\mu u^\nu &= 0 && \text{for light rays} \end{aligned} \quad \text{with} \quad u^\mu = \frac{dx^\mu}{ds}$$

where  $g_{\mu\nu}$  is the pseudo-Riemannian space-time metric

- reading of clocks = proper time of massive particles

$$s = \int_{\text{orbit}} ds = \int_{\text{orbit}} \sqrt{g_{\mu\nu} dx^\mu dx^\nu} = \int_{\text{orbit}} \sqrt{g_{\mu\nu} \dot{x}^\mu \dot{x}^\nu} dt$$





# Equations of motion: Extended bodies

Equation of motion for matter field

$$D_\nu T^{\mu\nu} = 0$$

is partial differential equation: difficult to solve

task: reduction of this PDE to a set of ordinary differential equations

→ Mathisson–Papapetrou–Dixon

# Equations of motion: Extended bodies

Mathisson–Papapetrou–Dixon: equation of particle with mass multipole and spin

$$\begin{aligned}
 v^\mu &= \frac{dx^\mu}{ds} \\
 D_v p_\mu &= R_{\mu\nu\rho\sigma} v^\nu S^{\rho\sigma} + D_\mu R_{\rho\sigma\tau\kappa} J^{\rho\sigma\tau\kappa} \\
 D_v S^{\mu\nu} &= v^\mu p^\nu - v^\nu p^\mu \\
 S^{\mu\nu} p_\nu &= 0
 \end{aligned}$$

## Quantities involved

- $v^\mu$  gives the geometric trajectory of the body
- $p_\mu$  is an auxiliary quantity describing the momentum (if  $p_\mu$  is considered as derived from  $T^\mu{}_\nu$ , then  $p_\mu$  is the primary quantity and  $v^\mu$  is an auxiliary quantity which however possesses the same interpretation as geometric orbit)
- $S^{\mu\nu}$  spin of particle
- $J^{\rho\sigma\tau\kappa\dots}$  mass multipole moments  
 special case: standard mass quadrupole  $J^{\rho\sigma\tau\kappa} = -3p^{[\rho} Q^{\sigma][\tau} p^{\kappa]}$  with  $Q^{\mu\nu} p_\nu = 0$



# Equations of motion: Extended bodies

Mathisson–Papapetrou–Dixon: equation of particle with mass multipole and spin

$$\begin{aligned}
 v^\mu &= \frac{dx^\mu}{ds} \\
 D_v p_\mu &= R_{\mu\nu\rho\sigma} v^\nu S^{\rho\sigma} + D_\mu R_{\rho\sigma\tau\kappa} J^{\rho\sigma\tau\kappa} \\
 D_v S^{\mu\nu} &= v^\mu p^\nu - v^\nu p^\mu \\
 S^{\mu\nu} p_\nu &= 0
 \end{aligned}$$

## Meaning

- direct access to curvature
- quadrupole motion in quadrupole space–time
  - special case: aligned quadrupoles
  - equatorial orbit possible: analytic solution?
  - exact quadrupole–quadrupole interaction in GR

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Plebański–Demiański space-time

stationary axially symmetric metric

$$ds^2 = \frac{\Delta_r}{p^2} (dt - A_\vartheta d\varphi)^2 - \frac{p^2}{\Delta_r} dr^2 - \frac{\Delta_\vartheta}{p^2} \sin^2 \vartheta (adt - A_r d\varphi)^2 - \frac{p^2}{\Delta_\vartheta} d\vartheta^2$$

where

$$p^2 = r^2 + (n - a \cos \vartheta)^2$$

$$\Delta_\vartheta = 1 + \frac{1}{3} a^2 \Lambda \cos^2 \vartheta - \frac{4}{3} \Lambda a n \cos \vartheta$$

$$\Delta_r = \left(1 - \frac{1}{3} \Lambda r^2\right) (r^2 + a^2) - 2Mr - n^2 + Q_e^2 + Q_m^2 - \Lambda n^2 (2r^2 + a^2 - n^2)$$

$$A_\vartheta = a \sin^2 \vartheta + 2n \cos \vartheta$$

$$A_r = r^2 + a^2 + n^2$$

- $M$  = mass,  $a$  = Kerr parameter,  $\Lambda$  = cosmological constant,  $n$  = NUT parameter,  $Q_e$  = electric charge,  $Q_m$  = magnetic charge
- this metric contains all standard black hole space-times, Petrov Type D
- Plebański & Demiański, AP 1976; Griffiths & Podolski, IJMP 2006
- horizons given by  $\Delta_r = 0$



# Conservation laws

There are two Killing vectors  $\partial_t$  and  $\partial_\varphi$

⇒ two conservation laws

$$E := g_{tt}\dot{t} + g_{t\varphi}\dot{\varphi}$$

$$-L := g_{\varphi t}\dot{t} + g_{\varphi\varphi}\dot{\varphi}$$

or

$$E = \frac{\Delta_r}{p^2}(\dot{t} - A_\varphi\dot{\varphi}) - a\frac{\Delta_\vartheta}{p^2}\sin^2\vartheta(a\dot{t} - A_r\dot{\varphi})$$

$$L = A_\vartheta\frac{\Delta_r}{p^2}(\dot{t} - A_\varphi\dot{\varphi}) - A_r\frac{\Delta_\vartheta}{p^2}\sin^2\vartheta(a\dot{t} - A_r\dot{\varphi}),$$

this corresponds to

- energy
- angular momentum in  $z$ -direction

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 **Space-times**
  - Vacuum space-times: Plebański–Demiański space-time
  - **String space-times**
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem



# String space-times

Plebański–Demiański string space-time  $d\varphi \rightarrow \beta d\varphi$

$$ds^2 = \frac{\Delta_r}{p^2} (dt - A_\vartheta \beta d\varphi)^2 - \frac{p^2}{\Delta_r} dr^2 - \frac{\Delta_\vartheta}{p^2} \sin^2 \vartheta (adt - A_r \beta d\varphi)^2 - \frac{p^2}{\Delta_\vartheta} d\vartheta^2$$

describes space-time with string along symmetry axis (space-time with matter)

- same Killing vectors
- conserved energy and angular momentum

→ see [Betti's talk](#)

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 **Space-times**
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - **Higher dimensions**
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# General Relativity in higher dimensions

spherically symmetric space-times in  $d$  dimensions: Reissner–Nordström–(anti-)de Sitter

$$ds^2 = g_{tt}dt^2 - g_{rr}dr^2 - r^2d\Omega_{d-2}^2$$

with

$$g_{tt} = \frac{1}{g_{rr}} = 1 - \frac{M^{d-3}}{r^{d-3}} + \frac{q^{2(d-3)}}{r^{2(d-3)}} - \frac{2\Lambda}{(d-1)(d-2)}r^2$$

Equation of motion

$$\begin{aligned} \left(\frac{dr}{d\varphi}\right)^2 &= \frac{r^4}{L^2} \frac{1}{g_{rr}g_{tt}} \left( E^2 - g_{tt} \left( \epsilon + \frac{L^2}{r^2} \right) \right) \\ &= \frac{r^4}{L^2} \left( E^2 - \left( 1 - \frac{M^{d-3}}{r^{d-3}} + \frac{q^{2(d-3)}}{r^{2(d-3)}} - \frac{2\Lambda}{(d-1)(d-2)}r^2 \right) \left( \epsilon + \frac{L^2}{r^2} \right) \right) \end{aligned}$$

Substitution  $u = \frac{m}{r}$ ,

various cases can be solved by elliptic and hyperelliptic integrals

→ for general case, see Valeria's talk



# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 **Space-times**
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - **PPN space-times**
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# PPN space-times

## PPN metric

$$g_{00} = 1 - 2U + 2\beta U^2 + \dots$$

$$g_{0i} = 0$$

$$g_{ij} = -(1 + 2\gamma U)\delta_{ij}$$

with Newtonian potential

$$U(t, \mathbf{x}) = \int \frac{\rho(t, \mathbf{x}')}{|\mathbf{x} - \mathbf{x}'|} d^3x'$$

- same Killing vectors
- conserved energy and angular momentum
- geodesic equation leads to differential equations which have the same mathematical structure as in Schwarzschild space-time

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - **The solutions**
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# The geodesic equation

geodesic equation

$$0 = \frac{d^2 x^\mu}{ds^2} + \left\{ \begin{matrix} \mu \\ \rho\sigma \end{matrix} \right\} \frac{dx^\rho}{ds} \frac{dx^\sigma}{ds}$$

is equivalent to the Hamilton–Jacobi equation

$$2 \frac{\partial S}{\partial s} = g^{\mu\nu} \frac{\partial S}{\partial x^\mu} \frac{\partial S}{\partial x^\nu}$$

separation ansatz

$$S = \frac{1}{2} \epsilon s - Et + L\varphi + S_r(r) + S_\vartheta(\vartheta)$$

- insertion into Hamilton–Jacobi
- separation of  $r$  and  $\vartheta$  equations
- separation constant =  $k$  = Carter constant (Carter, PR 1968)
- introduction of Mino time  $\tau$  through  $d\tau = \rho^2 ds$  (Mino, PRD 2003)
- substitution  $\xi = \cos \vartheta$
- renormalization: all quantities in units of  $r_S$



# The geodesic equation

$$\left(\frac{dr}{d\tau}\right)^2 = ((r^2 + a^2 + n^2)E - aL)^2 - \Delta_r(\epsilon r^2 + k) \quad =: R(r)$$

$$\left(\frac{d\xi}{d\tau}\right)^2 = \Delta_\xi(1 - \xi^2)(k - \epsilon(n - a\xi)^2) - (L - A_\xi E)^2 \quad =: \Theta(\xi)$$

$$\frac{d\varphi}{d\tau} = a \frac{(r^2 + a^2 + n^2)E - aL}{\Delta_r} + \frac{L - A_\xi E}{\Delta_\xi(1 - \xi^2)} \quad =: f(r) + g(\xi)$$

$$\frac{dt}{d\tau} = A_r \frac{(r^2 + a^2 + n^2)E - aL}{\Delta_r} + \frac{A_\xi(L - A_\xi E)}{\Delta_\xi(1 - \xi^2)} \quad =: h(r) + j(\xi)$$

analytic solution given by hyperelliptic functions (**Hackmann & C.L., PRL 2008**)

$$r(\tau) = \mp \frac{\sigma_2^{(r)}(\vec{x})}{\sigma_1^{(r)}(\vec{x})} + r_0 \quad \text{with} \quad \sigma^{(r)}(\vec{x}) = 0, \quad \vec{x} = \begin{pmatrix} \tau_1 \\ \tau \end{pmatrix}$$

$$\xi(\tau) = \mp \frac{\sigma_2^{(\xi)}(\vec{y})}{\sigma_1^{(\xi)}(\vec{y})} + \xi_0 \quad \text{with} \quad \sigma^{(\xi)}(\vec{y}) = 0, \quad \vec{y} = \begin{pmatrix} \tau_1 \\ \tau \end{pmatrix}$$



# The geodesic equation

integration of  $\varphi$  and  $t$  motion

$$\varphi - \varphi_0 = \int_{r_0}^{r(\tau)} f(r) \frac{dr}{\sqrt{R}} + \int_{\xi_0}^{\xi(\tau)} g(\xi) \frac{d\xi}{\sqrt{\Theta(\xi)}}$$

$$t - t_0 = \int_{r_0}^{r(\tau)} h(r) \frac{dr}{\sqrt{R}} + \int_{\xi_0}^{\xi(\tau)} j(\xi) \frac{d\xi}{\sqrt{\Theta(\xi)}}$$

$f$ ,  $g$ ,  $h$ , and  $j$  are rational functions

→ partial fraction expansion: hyperelliptic integrals of first, second and third kind

$$\int \frac{x^p dx}{\sqrt{P_n(x)}}, \quad \int \frac{x^q dx}{\sqrt{P_n(x)}}, \quad \int \frac{dx}{(x-c)\sqrt{P_n(x)}}$$

with  $p < \left[\frac{n-1}{2}\right]$  and  $q \geq \left[\frac{n-1}{2}\right]$ , can be integrated explicitly, but gives rather complicated expressions

regularity of geodesic equation for  $\vartheta = 0$  or  $\pi$



# The geodesic equation

solution for  $\varphi$  (similar for  $t$ )

$$\begin{aligned} \varphi = & \varphi_0 + \text{sign}(r'_0) (C_1^r f^r(\tau - \tau_0^r) + C_2^r(\tau - \tau_0^r) + I_{34}^r(\tau - \tau_0^r)) \\ & - \text{sign}(\vartheta'_0) (C_1^\vartheta f^\vartheta(\tau - \tau_0^\vartheta) + C_2^\vartheta(\tau - \tau_0^\vartheta) - aI_{32}^\vartheta(\tau - \tau_0^\vartheta) - I_{44}^\vartheta(\tau - \tau_0^\vartheta)) \end{aligned}$$

with

$$\begin{aligned} I_{mn}^x(w) = & \sum_{i=1}^n \frac{C_{m,i}^x}{\sqrt{P_5^x(u_i)}} \left( - \begin{pmatrix} f^x(w) - f^x(w_0) \\ w - w_0 \end{pmatrix}^T \cdot \int_{p_i^-}^{p_i^+} d\vec{r} \right. \\ & \left. + \frac{1}{2} \log \frac{\sigma \left( (f^x(w), w)^T - 2 \int_{\infty}^{p_i^+} d\vec{z} \right)}{\sigma \left( (f^x(w), w)^T - 2 \int_{\infty}^{p_i^-} d\vec{z} \right)} - (w \leftrightarrow w_0) \right) \end{aligned}$$

$d\vec{r}$  = holomorphic differentials of second kind

$d\vec{z}$  = meromorphic differentials of first kind

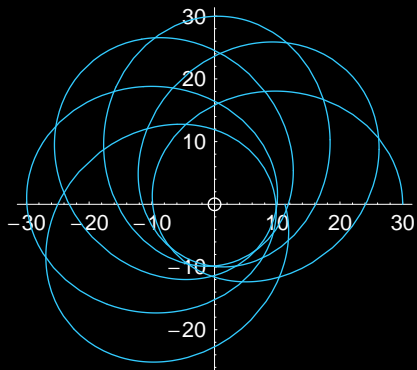
$$p_i^\pm = \left( \pm \frac{u_i}{\sqrt{P_5(u_i)}} \right) \quad \text{with} \quad u_i \text{ pole}$$



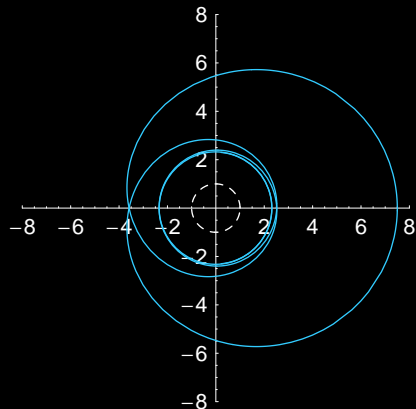
# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - **Examples for orbits**
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Schwarzschild



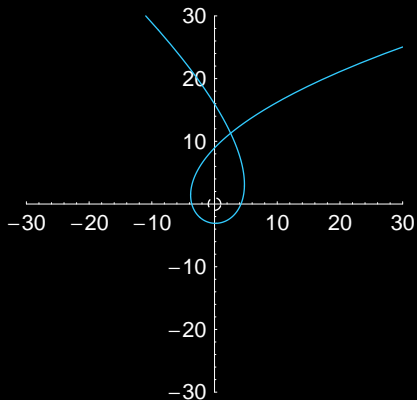
bound orbit



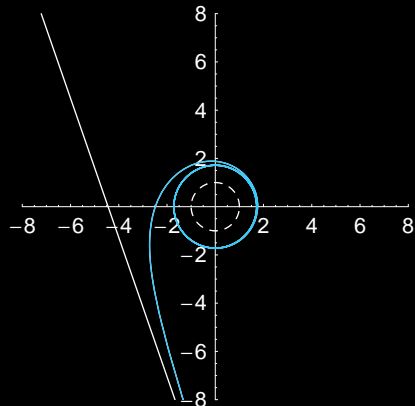
homocline orbit

Hagihara, JJGA 1931

## Schwarzschild



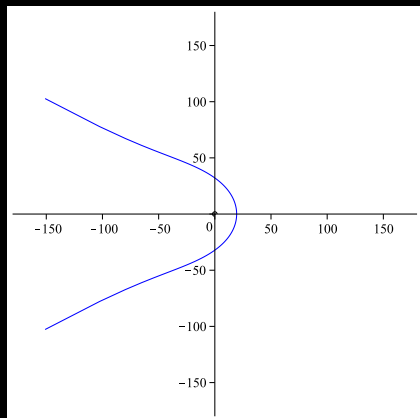
quasi parabolic escape orbit



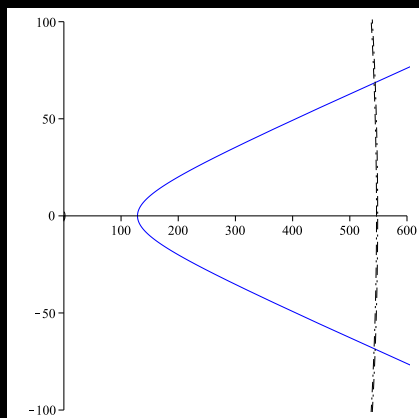
inspiral orbit

Hagihara, JJGA 1931

## Schwarzschild–de Sitter



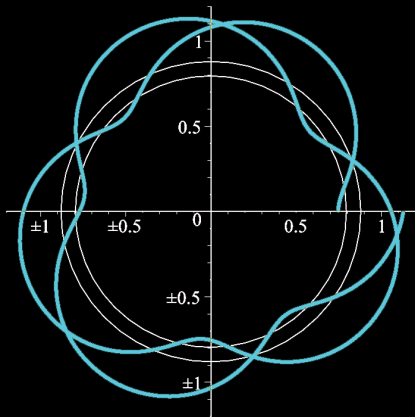
escape orbit



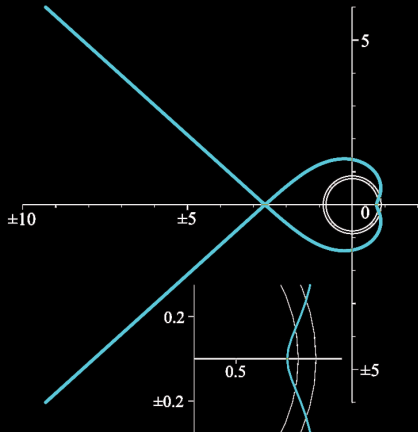
reflection at cosmic wall

Hackmann & C.L., PRL 2008, PRD 2008

# Reissner–Nordström in higher dimensions



bound orbit  
many universe orbit

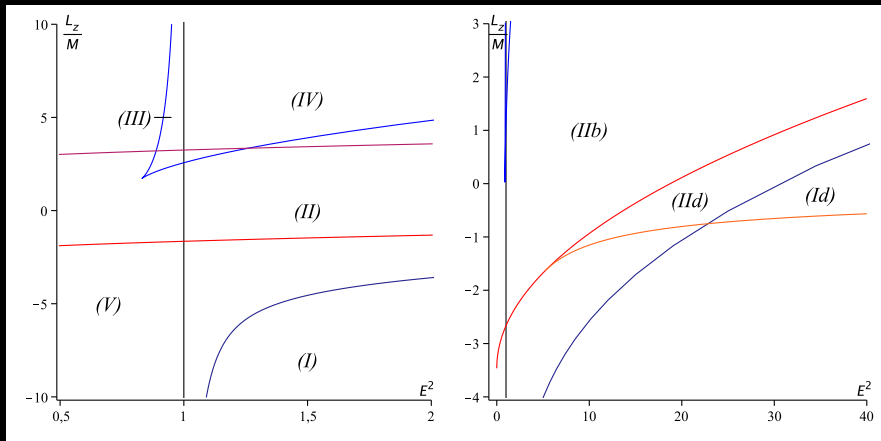


escape orbit  
escape in different universe

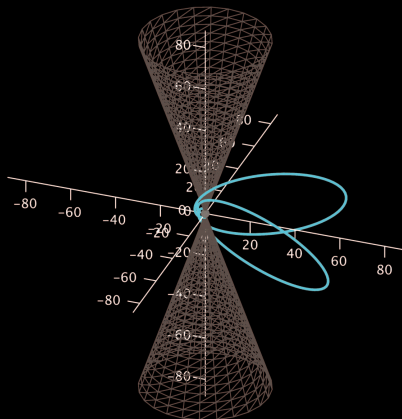
Hackmann, Kagramanova, Kunz, & C.L., PRD 2008



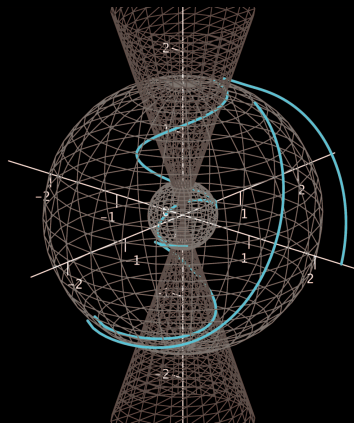
## Kerr space-time

parameter plots for  $r$  and  $\vartheta$ 

## Kerr-de Sitter



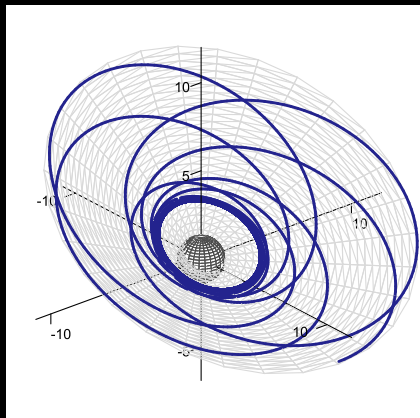
bound orbit



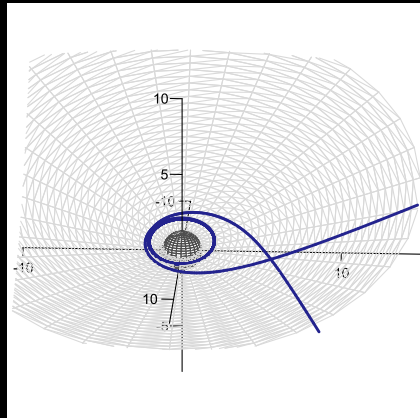
escape orbit

Hackmann, Kagramanova, Kunz, C.L., PRD 2010

## Taub-NUT



bound orbit



escape orbit

Hackmann, Kagramanova, Kunz, & C.L., PRD 2010

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Orbits

## Types of orbits

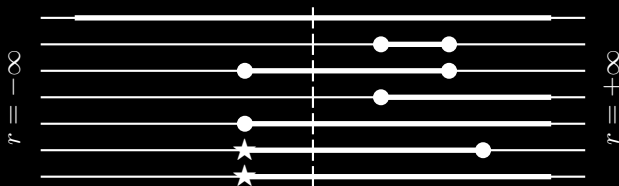
- bound orbit:  $r_{\min} \leq r \leq r_{\max}$
- escape orbit:  $r_{\min} \leq r \leq \infty$
- terminating bound orbit:  $r \leq r_{\max}$ , orbit terminates (falls into singularity)
- terminating escape orbit  $r \leq \infty$ , orbit terminates (falls into singularity)

$r$  can also become negative  $\Rightarrow$  more possibilities

# Orbits

## Types of orbits allowing $r < 0$

- transit orbit:  $-\infty < r < +\infty$
- bound orbit:  $0 < r_{\min} \leq r \leq r_{\max}$
- crossover bound orbit:  $r_{\min} \leq r \leq r_{\max}$  with  $r_{\min} < 0 < r_{\max}$
- escape orbit:  $r_{\min} \leq r \leq \infty$
- crossover escape orbit:  $r_{\min} \leq r \leq \infty$  with  $r_{\min} < 0$   
or  $-\infty < r < r_{\max}$  with  $0 < r_{\max}$
- terminating – bound – crossover orbit: bound orbit  $|r| < r_{\max}$  terminates
- terminating – escape – crossover orbit: escape orbit terminates



# Orbits

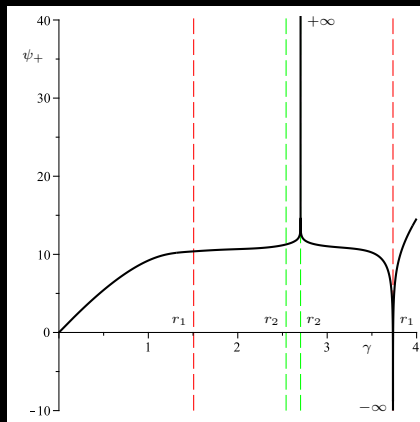
## Further discussion of effects

- geodesic incompleteness
- geodesics in analytic continuation of space-time
- closed time-like curves (CTC)
- crossing horizons
- homocline orbits
- many crosses of  $z$ -axes
- fast, slow rotation

# Taub–NUT space–time: incompleteness

- Taub–NUT space–time possess no curvature singularity
- but is geodesic incomplete ... during second transition through a horizon proper time terminates

(Hackmann, Kagramanova, Kunz, C.L. 2010)





# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 **Observables**
  - **Observables for bound orbits**
  - **Observables for escape orbits**
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 **Observables**
  - **Observables for bound orbits**
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Observables: for bound orbits

For bound orbits

- two oscillatory coordinates:  $r$  and  $\vartheta$  (generalized Lissajous figures)
- two (secularly) increasing coordinates:  $t$  and  $\varphi$

## Periods

- radial period

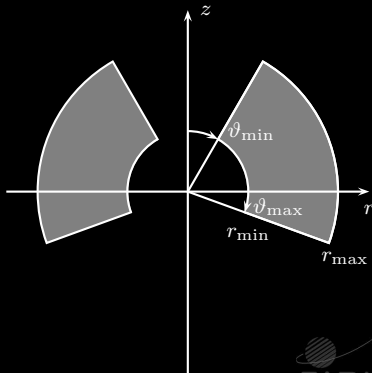
$$\omega_r = 2 \int_{r_{\min}}^{r_{\max}} \frac{dr}{\sqrt{R}}$$

is time needed to go from  $r_{\min}$  to  $r_{\min}$

- polar angle period

$$\omega_{\vartheta} = 2 \int_{\vartheta_{\min}}^{\vartheta_{\max}} \frac{dr}{\sqrt{\Theta}}$$

is time needed to go from  $\vartheta_{\min}$  to  $\vartheta_{\max}$



# Observables: for bound orbits

## Secular increases

- secular time increase

$$\Gamma = \left\langle \frac{dt}{d\tau} \right\rangle = \frac{2}{\omega_r} \int_{r_{\min}}^{r_{\max}} h(r) \frac{dr}{\sqrt{R}} + \frac{2}{\omega_\vartheta} \int_{\vartheta_{\min}}^{\vartheta_{\max}} j(\vartheta) \frac{dr}{\sqrt{\Theta}}$$

- secular azimuthal increase

$$Y = \left\langle \frac{d\varphi}{d\tau} \right\rangle = \frac{2}{\omega_r} \int_{r_{\min}}^{r_{\max}} f(r) \frac{dr}{\sqrt{R}} + \frac{2}{\omega_\vartheta} \int_{\vartheta_{\min}}^{\vartheta_{\max}} g(\vartheta) \frac{dr}{\sqrt{\Theta}}$$

orbital frequencies ([Drasco & Hughes, PRD 2004](#); [Schmidt, CQG 2004](#))

$$\Omega_r = \frac{2\pi}{\Gamma\omega_r}, \quad \Omega_\vartheta = \frac{2\pi}{\Gamma\omega_\vartheta}, \quad \Omega_\varphi = \frac{Y}{\Gamma}$$

- angular velocity of  $r$ -oscillations
- angular velocity of  $\vartheta$ -oscillations
- secular angular velocity

# Observables: for bound orbits

observables: self referential comparison, invariant

## The observables

- periastron shift

$$\Delta_{\text{periastron}} := \Omega_{\varphi} - \Omega_r = \left( Y - \frac{2\pi}{\omega_r} \right) \frac{1}{\Gamma}$$

- Lense–Thirring effect

$$\Delta_{\text{Lense–Thirring}} := \Omega_{\varphi} - \Omega_{\vartheta} = \left( Y - \frac{2\pi}{\omega_{\vartheta}} \right) \frac{1}{\Gamma}$$

- $\Delta_{\text{periastron}}$  compares the  $\varphi$ -advance for  $r_{\min}$  with  $2\pi$   
→ in weak field motion of  $r_{\min}$  within orbital plane or orbital cone
- $\Delta_{\text{Lense–Thirring}}$  compares the  $\varphi$ -advance for  $\vartheta_{\min}$  with  $2\pi$   
→ in weak field precession of orbital plane or orbital cone

all observables for bound orbits should be functions of  $\Omega_r$ ,  $\Omega_{\vartheta}$ , and  $\Omega_{\varphi}$  which can be evaluated explicitly by complete hyperelliptic integrals

# Observables: for bound orbits

- Schwarzschild, Schwarzschild–de Sitter, Reissner–Nordström, Taub–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} = 0$$

- Kerr, Kerr–de Sitter, Kerr–Newman, Kerr–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} \neq 0$$

# Observables: for bound orbits

- Schwarzschild, Schwarzschild–de Sitter, Reissner–Nordström, Taub–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} = 0$$

- Kerr, Kerr–de Sitter, Kerr–Newman, Kerr–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} \neq 0$$

- task: expansion of  $\Delta_{\text{perihelion}} = \Delta_{\text{perihelion}}(M, a, n, \Lambda, Q_e, Q_m)$  and  $\Delta_{\text{Lense–Thirring}} = \Delta_{\text{Lense–Thirring}}(M, a, n, \Lambda, Q_e, Q_m)$  (Hackmann, Kagramanova, Kunz, C.L., in preparation)

# Observables: for bound orbits

- Schwarzschild, Schwarzschild–de Sitter, Reissner–Nordström, Taub–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} = 0$$

- Kerr, Kerr–de Sitter, Kerr–Newman, Kerr–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} \neq 0$$

- task: expansion of  $\Delta_{\text{perihelion}} = \Delta_{\text{perihelion}}(M, a, n, \Lambda, Q_e, Q_m)$  and  $\Delta_{\text{Lense–Thirring}} = \Delta_{\text{Lense–Thirring}}(M, a, n, \Lambda, Q_e, Q_m)$  (Hackmann, Kagramanova, Kunz, C.L., in preparation)
- In general, for complicated potentials there are several periods  
 $\Rightarrow$  many perihelion shifts or Lense–Thirring effects ( $\rightarrow$  Valeria's talk)



# Observables: for bound orbits

- Schwarzschild, Schwarzschild–de Sitter, Reissner–Nordström, Taub–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} = 0$$

- Kerr, Kerr–de Sitter, Kerr–Newman, Kerr–NUT:

$$\Delta_{\text{perihelion}} \neq 0, \quad \Delta_{\text{Lense–Thirring}} \neq 0$$

- task: expansion of  $\Delta_{\text{perihelion}} = \Delta_{\text{perihelion}}(M, a, n, \Lambda, Q_e, Q_m)$  and  $\Delta_{\text{Lense–Thirring}} = \Delta_{\text{Lense–Thirring}}(M, a, n, \Lambda, Q_e, Q_m)$  (Hackmann, Kagramanova, Kunz, C.L., in preparation)
- In general, for complicated potentials there are several periods  $\Rightarrow$  many perihelion shifts or Lense–Thirring effects ( $\rightarrow$  Valeria's talk)
- observables on higher dimensions: many  $\varphi_p, \vartheta_p, p = 1, \dots, n - 3$  periods possible to define  $\Gamma$  and  $Y_p$ , and

$$\begin{aligned} \Delta_{\text{periastron},p} &:= \Omega_{\varphi_p} - \Omega_p \\ \Delta_{\text{Lense–Thirring},p,p'} &:= \Omega_{\varphi_p} - \Omega_{\vartheta_{p'}} \end{aligned}$$

does this make sense?



# Post-Schwarzschild of perihelion shift

Perihelion shift in Schwarzschild–de Sitter (for bound orbit, **Kraniotis & Whitehouse, CQG 2003**)

$$\delta\varphi_{\text{perihelion}} = 2\pi - \omega_{22} = 2\pi - \oint \frac{x dx}{\sqrt{P_5(x)}}$$

with

$$\begin{aligned} \oint_{a_2} \frac{x dx}{\sqrt{P_5(x)}} &= \oint_x \frac{1}{\sqrt{P_3(x)}} - \frac{2}{3} \Lambda m^2 \oint_{a_2} \frac{x^2 + \lambda}{x^2 P_3(x) \sqrt{P_3(x)}} dx + \mathcal{O}(\Lambda^2) \\ &= \omega_1 + \Lambda \frac{m^2}{96} \left( \sum_{j=1}^3 \frac{\eta_1 + \omega_1 z_j}{(\varphi''(\rho_j))^2} \left( 1 + \frac{\lambda}{(4z_j + \frac{1}{3})^2} \right) \right. \\ &\quad \left. + \lambda \left( \frac{2\eta_1 - \frac{1}{6}\omega_1}{16(\varphi'(u_0))^2} + \frac{6}{16} \frac{\varphi''(u_0)}{(\varphi'(u_0))^5} (\zeta(u_0) - \eta_1 u_0) \right) \right) + \mathcal{O}(\Lambda^2) \end{aligned}$$

- Needs introduction of  $r_{\min}$  and  $r_{\max}$  or  $a$  and  $e$  for interpretation
- Needs relativistic approximation for interpretation

# Perihelion shift

$$\omega_1 = \int_{r_{\min}}^{r_{\max}} \frac{d\varphi}{dr} dr = \int_{e_2}^{e_3} \frac{d\varphi}{dx} dx = \int_{e_2}^{e_3} \frac{dx}{\sqrt{\left(\frac{dx}{d\varphi}\right)^2}} = \int_{e_2}^{e_3} \frac{dx}{\sqrt{4x^3 - g_2x - g_3}}$$

Perihelion shift

$$\delta\varphi = 2\omega_1 - 2\pi = \frac{4}{\sqrt{-e_2 - 2e_3}} \int_0^{\frac{\pi}{2}} \frac{dx}{\sqrt{1 - \frac{e_2 - e_3}{-e_2 - 2e_3} \sin^2 x}} - 2\pi.$$

One can identify

$$e_2 = \frac{2M}{r_{\min}} - \frac{1}{3}, \quad e_3 = \frac{2M}{r_{\max}} - \frac{1}{3}.$$

- Can be used for approximation
- Can be used for representation in terms of semi-major axis and eccentricity

# Observables: for bound orbits

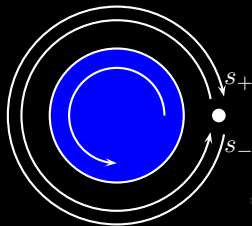
## Further observables

- progression of nodes: determination of  $\varphi$  and  $t$  or  $\tau$  for which  $\vartheta = \pi/2$ : then

$$\Delta_i \varphi = \varphi(\tau_{i+1}) - \varphi(\tau_i) \quad \text{with} \quad \vartheta(\tau_i) = \frac{\pi}{2}, \quad i = 1, 2, \dots$$

one has to determine  $\tau_i$  and then to integrate  $d\varphi/d\tau$  from  $\tau_i$  to  $\tau_{i+1}$   
(Gebhardt, Hackmann & C.L., in preparation)

- clock effect  $s_+ - s_- \sim 4\pi \frac{J}{M} \sim 10^{-7} \text{ s}$  — and generalizations of it ...
- analytic expressions still have to be calculated
- application to Galileo?



# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables**
  - Observables for bound orbits
  - **Observables for escape orbits**
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Observables: for flyby orbits

flyby orbit:  $r \rightarrow \infty$  for  $s \rightarrow \pm\infty$

Then  $\vartheta^\pm = \vartheta(\pm\infty)$  and  $\varphi^\pm = \varphi(\pm\infty)$

## Deflection angles

- azimuthal deflection angle

$$\Delta\varphi = \varphi^+ - \varphi^-$$

- polar deflection angle

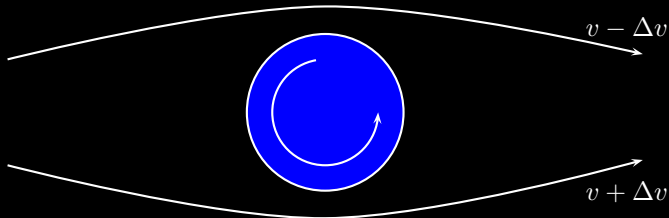
$$\Delta\vartheta = \vartheta^+ - \vartheta^-$$

- analytic expressions still have to be calculated
- application to rotating black holes
- no impact on flyby anomaly ([Hackmann & C.L. 2010](#))

In addition for light:

- gravitational time delay

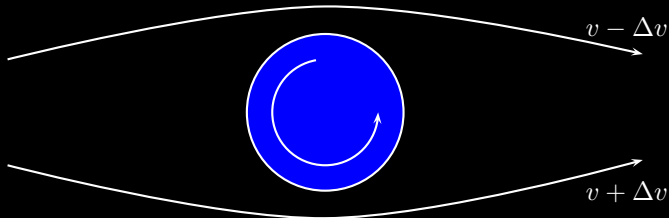
# Further observables



- flyby at rotating body: different direction, different velocity
- no impact on flyby anomaly ([Hackmann & C.L. 2010](#))
- deflection of light
- timing formula

will depend on impact parameter as well as on polar angle

# Further observables



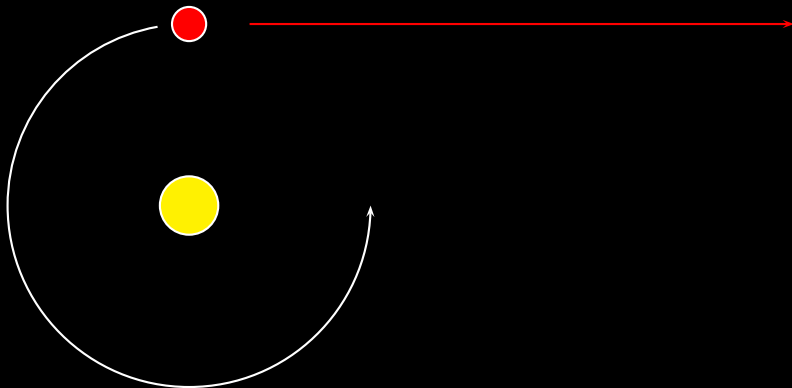
- flyby at rotating body: different direction, different velocity
- no impact on flyby anomaly (**Hackmann & C.L. 2010**)
- deflection of light
- **timing formula**

will depend on impact parameter as well as on polar angle



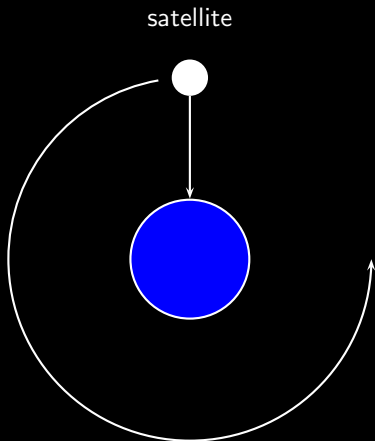
# Timing formula

pulsar



Question: frequency of pulses arriving at Earth — function of orbit (position, velocity, structure of gravitational field)

# Timing formula



Question: frequency of satellites  
(time of satellites, Galileo) arriving at  
Earth — function of orbit

- signals arriving at fixed position  
on Earth
- signals arriving at surface of  
Earth

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem

# Discussion: General meaning

## Theorem: Separability

The Hamilton–Jacobi equation for the geodesic equation is separable if and only if space–time is of Petrov type D without acceleration ([Demianski & Francaviglia, JPA 1981](#))

## Theorem: Petrov type D

The general type D Petrov space–times are exhausted by the electro–vac Plebański–Demiański solutions ([Plebański & Demiański, AP 1976](#))

## Theorem: Integration

The geodesic equation in a general electro–vac Plebański–Demiański space–time without acceleration can be integrated using the method of hyperelliptic integrals ([Hackmann, Kagramanova, Kunz, C.L., EPL 2009](#))

All what is separable can be solved analytically

# Summary and outlook

## Summary

- Complete analytic solution of geodesic equation in Plebański–Demiański space-times
- Analytic solution for all electro–vac space-times for which Hamilton–Jacobi separates
- Complete set of fundamental observables for bound orbits

mathematics is essentially under control → discussion of solutions and observables

# Summary and outlook

## Outlook 1: discussion of obtained solutions

- analytic description of progression of nodes
- further observables (deflection angle, clock effect, timing, time delay, ...)
- post-Newton, post-Schwarzschild, post-Kerr, ... expansions of solutions
- post-Newton, post-Schwarzschild, post-Kerr, ... expansions of observables

## Outlook 2: new solutions

- motion in axially symmetric mass multipole fields (e.g. [Quevedo, FP 1990](#))
- we are now able to analytically solve

$$\left(\frac{dr}{d\tau}\right)^2 = P_n(r) \quad \text{for all } n$$

(Enolskii, Hackmann, Kagramanova, Kunz, C.L. in preparation, → [see Valeria's talk](#))

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - Effective one-body problem



# Spinning objects

spherically symmetric metric

$$ds^2 = \alpha dt^2 - \alpha^{-1} dr^2 + r^2 d\vartheta^2 + r^2 \sin^2 \vartheta d\varphi^2, \quad \alpha = 1 - \frac{2M}{r} - \frac{\Lambda r^2}{3} + \frac{Q^2}{r^2}$$

additional constant of motion

$$C \equiv \xi^\mu p_\mu + D_\nu \xi_\mu S^{\mu\nu}$$

spin motion can be solved

$$\frac{dS^{\mu\nu}}{ds} - \frac{1}{r} \frac{dr}{ds} S^{r\varphi} = 0 \quad \Rightarrow \quad S^{r\varphi} = \frac{S}{r}$$

equation of motion for radial coordinate (with  $J = L + S$ )

$$\left( \frac{dr}{ds} \right)^2 = E^2 - \frac{(\partial_r g_{tt}) S J}{r} + g_{\varphi\varphi} g_{tt} \left( \frac{J^2 + 2JS}{r^4} \right) - g_{tt}$$

# Spinning objects

for Reissner–Nordström–de Sitter

$$\left(\frac{dr}{ds}\right)^2 = E^2 - 1 + \frac{\Lambda r^2}{3} - \Lambda S^2 - \frac{2\Lambda LS}{3} + \frac{\Lambda L^2}{3} + \frac{2}{r} + \frac{S^2 - L^2 - Q^2}{r^2} + \frac{2LS + 2L^2}{r^3} - Q^2 \frac{(L + S)^2}{r^4}$$

# Outline

- 1 Introduction and motivation
- 2 General Relativity
- 3 Space-times
  - Vacuum space-times: Plebański–Demiański space-time
  - String space-times
  - Higher dimensions
  - PPN space-times
- 4 Solutions of the geodesic equation
  - The solutions
  - Examples for orbits
  - Classification of orbits
- 5 Observables
  - Observables for bound orbits
  - Observables for escape orbits
- 6 Discussion, summary and outlook
  - Discussion
  - The solution for spinning objects
  - **Effective one-body problem**

# Effective one-body problem

- Newton: 2-body problem can be reduced to a one-body problem
- Einstein: not possible in closed form
- series expansion method for successive reduction to an “effective” one-body problem within post-Newtonian expansion
- effective dynamics of two black holes described by

$$ds^2 c = -g_{tt}(r, \nu) dt^2 + g_{rr}(r, \nu) dr^2 + r^2 (d\vartheta^2 + \sin^2 \vartheta d\varphi^2)$$

with  $u = 2(M_1 + M_2)/r$ ,  $\nu = M_1 M_2 / (M_1 + M_2)^2$  and

$$\begin{aligned} g_{tt}(r, \nu) &= 1 - 2u + 2\nu u^3 + \nu a_4 u^4 + \mathcal{O}(u^5) \\ (g_{tt}(r, \nu) g_{rr}(r, \nu))^{-1} &= 1 + 6\nu u^2 + 2(26 - 3\nu)\nu u^3 + \mathcal{O}(u^4) \end{aligned}$$

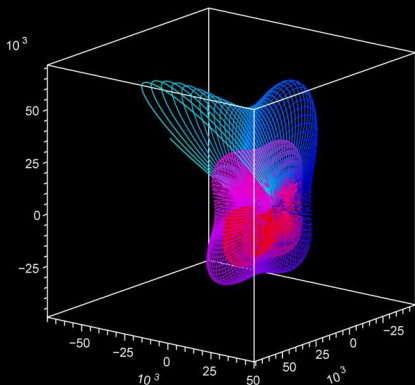
- effective one-body equation of motion

$$\left( \frac{dr}{d\varphi} \right)^2 = \frac{r^4}{L^2} \frac{1}{g_{rr} g_{tt}} \left( E^2 - g_{tt} \left( \epsilon + \frac{L^2}{r^2} \right) \right)$$

# Effective one-body problem

- physical questions
  - orbits
  - last stable orbit
  - last spherical stable orbit
  - last circular stable orbit
- has to be complemented by radiation reaction effects
  - variation of orbital parameters, variation of observables
  - gravitational radiation
  - inspiraling orbit
- can be supplemented by spin  $\rightarrow$  axially symmetric case

# The End



# Thank you

Also thanks to

- German Research Foundation DFG
- Center of Excellence QUEST
- German Space Agency DLR